

ABSTRACT

A cyclic swashplate device for controlling rotorcraft blade pitch is disclosed for application to rotorcraft swashplates, in particular, in helicopters. A cyclic swashplate device controls the rotorcraft blade pitch. The device (10'), with rotating (10') and non-rotating (14') cyclic swashplates is designed in such a way that at least one of the two disks includes a modular link fitting assembly (46, 42) ensuring the links with the disk (12', 14') and the pitch connecting rods (6) and/or at least one driving device or with the pilot control devices (17) and/or at least one retaining device. Interconnecting fittings are attached rigidly and separately to an annular device, such as one of the rings (31', 30') of a bearing (21') on the corresponding disk (14', 12').